

1 Sub-meter level navigation with an enhanced multi-GNSS single point positioning algorithm
2 using iGMAS ultra-rapid products

3
4 Sinan Birinci ¹, Mehmet Halis Saka ^{1,*}

5 ¹ Gebze Technical University, Department of Geomatics Engineering, 41400 Gebze-Kocaeli,
6 Turkey

7 s.birinci@gtu.edu.tr , saka@gtu.edu.tr

8
9 ORCID iDs

10 Sinan Birinci <https://orcid.org/0000-0001-6668-9160>

11 Mehmet Halis Saka <https://orcid.org/0000-0002-5283-9065>

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34 Abstract

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36 The ultra-rapid products have the advantage of being used in real-time positioning with no
37 external connections. In this study, these products provided by the international GNSS
38 Monitoring and Assessment System (iGMAS) for four global constellations (GPS, GLONASS,
39 Galileo, and BDS-3) were assessed in terms of service rate and accuracy in navigation. In this
40 regard, a Matlab-based in-house code solving the problem was developed for all possible
41 combinations of the constellations. To explore the effectiveness of iGMAS products, the same
42 dataset has been also processed using the GFZ rapid products. The results demonstrate that GPS
43 and Galileo solutions were substantially comparable to the rapid products concerning service
44 rate and accuracy, but GLONASS and BDS-3 iGMAS products require some enhancements. In
45 addition, Galileo produced remarkably good results both individually and combinational. The
46 GPS/GLONASS/Galileo/BDS-3 SPP solution generated a mean RMS error of 0.54 m
47 horizontally and 0.89 m vertically. Thus, GPS-only, GLONASS-only, Galileo-only, and BDS-
48 3-only solutions were improved by 42%, 79%, 28%, and 74% in 3D mean RMS error with the
49 quad system solutions, respectively.

50

51 Keywords: Precise navigation, iGMAS ultra-rapid products, multi-GNSS SPP

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53

54 1. Introduction

55

56 In the wake of GPS, operational usage of Global Navigation Satellite Systems (GNSS)
57 technology, together with recently developed projects, such as GLONASS, Galileo, and the
58 BeiDou Satellite System (BDS) has become more powerful, and reliable in satellite-based
59 positioning (Li et al., 2015). Especially after 2010, modernizations and innovations in these
60 systems rendered each of them operative in worldwide use. In the renovation stage, the revisions
61 made in the Russian satellites system GLONASS, nominal 24 Medium Earth Orbit (MEO)
62 operational satellites have been in service since 2012 (Revnivykh et al., 2017). Today, the
63 constellation consists of 18 GLONASS-M satellites, 2 GLONASS-M+ satellites, and 1
64 GLONASS-K satellite (Bury et al., 2022). The Chinese BDS was initially realized as a
65 demonstration system BDS-1, followed by the regional BDS-2, which specifically serves the
66 Asia-Pacific region, was completed at the end of 2012. The third-generation navigation system
67 BDS-3, which aims to serve the whole world, was announced on 27 December 2018. Currently,

68 this system includes 24 MEO, 3 Geostationary Earth Orbit (GEO), and 3 Inclined
69 Geosynchronous Orbit (IGSO) satellites (Yuanxi Yang et al., 2019), (CSNO-TARC, 2022).
70 The Galileo constellation project was started by the European Union towards the end of 2005.
71 The success of the only-Galileo positioning solution was tested by launching two pairs of In-
72 Orbit Validation (IOV) satellites in 2011 and 2012. In 2014, the constellation was developed
73 with Full Operational Capability (FOC) satellites. As of February 2022, Galileo contains usable
74 22 MEO satellites (EUSPA, 2022). GPS, GLONASS, Galileo, and BDS-3 are the basic
75 components of Positioning Navigation and Timing (PNT) services capable to provide
76 information to the users independently.

77
78 Each constellation has a comparable structure, mathematical modeling, and observation type.
79 Therefore, an integrated positioning has become possible which is named multi-GNSS together
80 with the new generation receivers (Montenbruck et al., 2014). Obtaining precise orbit and clock
81 products of these systems has become a requirement for multi-GNSS Precise Point Positioning
82 (PPP) applications. International GNSS Service (IGS) began providing these products for GPS
83 satellites in 1994 (G. Beutler et al., 1999). The orbits of all operational GLONASS satellites
84 were included in the IGS precise products in 2004 (Gerhard Beutler et al., 2009). With the
85 advent of Galileo, BDS, and the other regional navigation systems within the framework of
86 GNSS, the IGS Multi-GNSS Experiment (MGEX) project was put on the agenda with the
87 modern receivers deployed in 2013. Thus, different Analysis Centers (ACs) have started to
88 calculate precise products utilizing MGEX network data for GPS, GLONASS, Galileo, and
89 BDS (Montenbruck et al., 2017). According to the publishing timeliness, IGS precise products
90 are classified as final, rapid, and ultra-rapid, with the final product having a delay of 12–18
91 days, the rapid product having a delay of 17 hours, and the ultra-rapid product (observed half)
92 having a delay of 3–9 hours, although another type of ultra-rapid product (predicted half) can
93 be acquired in real-time (Dow et al., 2009; Shi et al., 2015).

94
95 Thanks to the MGEX products, especially multi-GNSS PPP has become applicable. Therefore,
96 many PPP studies in the literature started using GPS-GLONASS and continued with the
97 integration of Galileo, and BDS systems (Cai and Gao, 2013; Jiao et al., 2019; Kiliszek and
98 Kroszczyński, 2020). Parallel to the post-processing applications, IGS Real-Time Service
99 (RTS) was officially initiated in 2013 which provides orbit and clock correction for the
100 broadcast ephemeris to the real-time PPP users (Hadaś and Bosy, 2015; Wang et al., 2018; Ge
101 et al., 2021). In order to acquire RTS orbit and clock corrections, which are broadcasted via an

102 internet stream, a strong external connection is required. During real-time positioning, an
103 unexpected communication failure or network connection latency will cause a gap of several
104 seconds to hours in receiving orbital and clock data. Hence, an alternate approach to the RTS
105 might be considered by utilizing ultra-rapid products to ensure positioning uninterruptedly (El-
106 Mowafy et al., 2017). The ultra-rapid products are used in a variety of studies, including near-
107 time troposphere estimation (Satirapod et al., 2011, Hadas et al., 2013), precise timing (Cerretto
108 et al., 2012), clock estimation (Cao et al., 2022), examination of dynamic displacements (Yigit
109 et al., 2020), earthquake analysis (Shu et al., 2020), time transfer (Chen et al., 2022), real-time
110 PPP and SPP applications (Elsobeiey and Al-Harbi, 2016; Bahadur and Nohutcu, 2021; Geng
111 et al., 2022; Jiao and Song, 2022; Jiang et al., 2022; Ogutcu and Farhan, 2022; Bahadur, 2022).
112 ~~The ultra rapid products are used in many studies, including near real time estimation of the~~
113 ~~troposphere (Satirapod et al., 2011; T. Hadas et al., 2013), precise timing (Cerretto et al., 2012),~~
114 ~~earthquake analysis (Shu et al., 2020), and evaluation of positioning performances (Elsobeiey~~
115 ~~and Al Harbi, 2016; Bahadur and Nohuteu, 2021; Oguteu and Farhan, 2022).~~

116
117 Nowadays, ultra-rapid products are estimated by both IGS and International GNSS Monitoring
118 and Assessment System (iGMAS) independently. IGS uses eight ACs in that can be listed as
119 Center for Orbit Determination in Europe (CODE), Center National d'Etudes Spatiales and
120 Collecte Localization Satellites (CNES/ CLS), Wuhan University (WHU), Deutsches
121 GeoForschungs Zentrum (GFZ), Japan Aerospace Exploration Agency (JAXA), Technische
122 Universität München (TUM), Shanghai Astronomical Observatory (SHAO), and Information
123 and Analysis Center (IAC) (Shen et al., 2021). In addition to the IGS products, the Chinese
124 project called iGMAS provides products obtained from combinations of its own ACs.
125 Currently, 12 iGMAS ACs process the multi-GNSS data collected from about 30 iGMAS
126 tracking stations and generate precise GNSS products. Each ACs has the same mission and
127 operates independently (iGMAS, 2022; W. Zhou et al., 2022). When comparing IGS and
128 iGMAS, the main difference is that they are based on two different networks. In addition,
129 iGMAS offers users an important opportunity to provide BDS ultra-rapid products with the
130 contribution of 12 ACs, whereas the IGS offers this only by WHU and CNES. Therefore, the
131 iGMAS ultra-rapid products availability for BDS looks stronger than IGS. iGMAS also
132 provides ultra-rapid clock products for four GNSS systems. Consequently, iGMAS ultra-rapid
133 products make positioning possible individually or in different combinations of GNSS systems
134 in real-time.

135

136 The current status of iGMAS products, particularly in the scope of BDS-3, directed us to use
 137 them in navigation problems. Hence, the Single Point Positioning (SPP) technique was applied
 138 to investigate the contribution of the real-time products to the navigation solution. For the
 139 processing of data, a comprehensive Matlab code was developed in this study. The dataset used
 140 was taken from sixteen IGS MGEX stations. A detailed analysis was performed in terms of
 141 positioning accuracy with single, dual, triple, and quad systems. The service rates provided by
 142 iGMAS products for GPS, GLONASS, Galileo, and BDS-3 were investigated. Furthermore,
 143 for evaluating the quality of iGMAS ultra-rapid products, GFZ rapid products were also
 144 employed, and their performance was compared with those obtained from the iGMAS products.

145
 146 This study is divided into 5 parts. In Section 2, the multi-GNSS SPP functional and stochastic
 147 model as well as the outlier detection technique are explained in detail. In Section 3, the
 148 introduction of the dataset and the evaluation strategy of the developed code is given. In Section
 149 4, an extensive accuracy analysis has been performed for all combinations with iGMAS
 150 products. Finally, Section 5 is summarized the conclusions obtained from the study.

151

152 2. Methodology

153

154 2.1. Multi-GNSS SPP model

155

156

157 Multi-GNSS SPP technique provides users with time and position information based on
 158 pseudorange observations from multiple satellite systems. The equation of pseudorange ($P_{i,r}^{s,j}$)
 159 measurements on *ith* ($i=1,2$) frequency can be written as follows:

160

$$P_{i,r}^{s,j} = \rho_r^{s,j} + cdT_r^s - cdT^{s,j} + Tr^{s,j} + I_i^{s,j} + b_{i,r}^s - b_i^{s,j} + \varepsilon_{P_{i,r}^{s,j}} \quad (1)$$

161

162 where superscript s represents the GNSS constellations (G : GPS, R : GLONASS, E : Galileo,
 163 C : BDS-3) and j denotes the evaluated satellite of each constellation. Subscript r and i refer
 164 to GNSS receiver and frequency of GNSS signal, respectively. $\rho_r^{s,j}$ is the geometric range in
 165 meters between the satellite and the receiver, c is the speed of light in a vacuum in meters per
 166 second, $cdT^{s,j}$ and cdT_r^s are satellite and receiver clock offsets in seconds, respectively. $Tr^{s,j}$
 167 is the tropospheric delay in meters, $I_i^{s,j}$ is the first order ionospheric delay depending on *ith*

168 frequency in meters; $b_{i,r}^s$ and $b_i^{s,j}$ are the receiver and satellite hardware code delay in
 169 pseudorange observations at i th frequency in meters; $\varepsilon_{P_{i,r}^{s,j}}$ consists of the measurement noise,
 170 multipath, orbital error, etc. in meters. It should be outlined that some systematic errors such as
 171 satellite phase offset, Earth rotation effects, relativistic effects, and solid tide effect were
 172 removed by using common models (Kouba and Héroux, 2001). The first-order ionospheric
 173 delay was eliminated by using ionosphere-free (*IF*) combination of the dual-frequency data.
 174

175 Precise products are provided by many ACs to remove satellite orbit and clock offset from the
 176 equation. The satellite clock offset products, which have a high correlation with the satellite
 177 hardware code delays, are generated by ionosphere-free code observations. Thereby, these
 178 estimated satellite clock offset products include the *IF* combination of satellite hardware code
 179 delays. Furthermore, the receiver hardware code bias which is strongly correlated with the
 180 receiver clock offset is absorbed by the receiver clock offset parameter in the estimation (Kouba
 181 and Héroux, 2001; F. Zhou et al., 2019). Thus, these two parameters were completely
 182 disregarded in the model during the evaluation process. This can be expressed with the
 183 following equations.

$$184 \quad \tilde{P}_{IF,r}^{s,j} = P_{IF,r}^{s,j} + cd\tilde{T}^{s,j} \quad (2)$$

$$185 \quad cd\tilde{T}^{s,j} = cdT^{s,j} + b_{IF}^{s,j} \quad (3)$$

$$186 \quad cd\tilde{t}_r^s = cd t_r^s + b_{IF,r}^s \quad (4)$$

187
 188 where $\tilde{P}_{IF,r}^{s,j}$, is the pseudorange observation corrected by the satellite clock offset. $cd\tilde{T}^{s,j}$ and
 189 $cd\tilde{t}_r^s$ are redefined satellite and receiver clock offset, respectively. When using multi-GNSS,
 190 each constellation has different hardware delays and time scales. For this reason, two techniques
 191 are available to evaluate receiver clock offset in multi-GNSS. The first one is to independently
 192 obtain the receiver clock offset for each constellation, which is preferred in this study. The
 193 second method is to estimate the receiver clock offset for a selected reference system and the
 194 Inter System Bias (ISB) parameters between the reference system and the others (Lou et al.,
 195 2016; Liu et al., 2019). For tropospheric delay, the UNB3m model, developed by the University

196 of New Brunswick, using the Niell Mapping Function and known as a hybrid model, was used
 197 in this study (Niell, 1996; Leandro et al., 2008). Using dual-frequency GNSS data, the IF
 198 pseudorange observations of the multi-constellation integration may be stated as follows.
 199

$$\begin{cases} \tilde{P}_{IF,r}^{G,j} = \rho_r^{G,j} + cd\tilde{t}_r^G + \varepsilon_{P_{IF,r}^{G,j}} \\ \tilde{P}_{IF,r}^{R,j} = \rho_r^{R,j} + cd\tilde{t}_r^R + \varepsilon_{P_{IF,r}^{R,j}} \\ \tilde{P}_{IF,r}^{E,j} = \rho_r^{E,j} + cd\tilde{t}_r^E + \varepsilon_{P_{IF,r}^{E,j}} \\ \tilde{P}_{IF,r}^{C,j} = \rho_r^{C,j} + cd\tilde{t}_r^C + \varepsilon_{P_{IF,r}^{C,j}} \end{cases} \quad (5)$$

200
 201 The observation model (Equation (5)) should be linearized around the approximate receiver
 202 position. For the estimation of receiver coordinates and clock parameters, the common Gauss-
 203 Markov model together with the weighted least squares method is used. The equations are given
 204 by (Koch, 1999),
 205

$$E(\mathbf{l}) = \mathbf{l} + \mathbf{v} = \mathbf{A}\mathbf{x} \quad ; \quad E(\mathbf{v}) = 0 \quad \text{and} \quad D(\mathbf{l}) = \sigma_0^2 \mathbf{Q}_l = \sigma_0^2 \mathbf{P}^{-1} \quad (6)$$

$$\mathbf{x} = (\mathbf{A}^T \mathbf{P} \mathbf{A})^{-1} \mathbf{A}^T \mathbf{P} \mathbf{l} \quad (7)$$

207
 208
 209 where $E(\mathbf{l})$ is the expected value of observations, $D(\mathbf{l})$ is the variance-covariance matrix of
 210 the observations, \mathbf{v} is the posterior residuals of measurement vector, σ_0^2 is the a priori variance
 211 factor, \mathbf{x} is the estimated unknown parameter vector, \mathbf{A} is the coefficients matrix of the
 212 linearized equation, \mathbf{l} is the measurement vector, which is the difference between the corrected
 213 pseudorange measurement and the distance calculated using the satellite coordinates and the
 214 receiver approximate coordinates, \mathbf{P} is the weight matrix of the observations, \mathbf{Q}_l is the cofactor
 215 matrix of observations. The multi-GNSS SPP observation model can be explicitly expressed
 216 as follows:
 217

$$\mathbf{v} = \mathbf{Ax} - \mathbf{l} = \begin{bmatrix} \mathbf{v}^G \\ \mathbf{v}^R \\ \mathbf{v}^E \\ \mathbf{v}^C \end{bmatrix} = \begin{bmatrix} \mathbf{A}^G & \mathbf{B}^G & 0 & 0 & 0 \\ \mathbf{A}^R & 0 & \mathbf{B}^R & 0 & 0 \\ \mathbf{A}^E & 0 & 0 & \mathbf{B}^E & 0 \\ \mathbf{A}^C & 0 & 0 & 0 & \mathbf{B}^C \end{bmatrix} \begin{bmatrix} \Delta \mathbf{r} \\ cdt^G \\ cdt^R \\ cdt^E \\ cdt^C \end{bmatrix} - \begin{bmatrix} \mathbf{l}^G \\ \mathbf{l}^R \\ \mathbf{l}^E \\ \mathbf{l}^C \end{bmatrix} \quad (8)$$

218
 219 where $\Delta \mathbf{r} = [\Delta x \ \Delta y \ \Delta z]^T$ parameters vector for receiver coordinates, \mathbf{B} is the clock offsets
 220 coefficients vector in that each element equals one. $\mathbf{A}^G, \mathbf{A}^R, \mathbf{A}^E, \mathbf{A}^C$ in the model are sub-
 221 coefficients matrices of coordinate unknowns for different GNSS. These sub-matrices can be
 222 expressed for GPS observations, as follows:

$$[\mathbf{v}^G] = [\mathbf{A}^G \ \mathbf{B}^G] \begin{bmatrix} \Delta \mathbf{r} \\ cdt^G \end{bmatrix} - [\mathbf{l}^G] \quad (9)$$

$$\mathbf{A}^G = \begin{bmatrix} \mathbf{a}_1 \\ \mathbf{a}_2 \\ \vdots \\ \mathbf{a}_n \end{bmatrix}; \quad \mathbf{a}_i = \begin{bmatrix} \frac{-(X^i - X_r)}{\|\mathbf{r}^s - \mathbf{r}_r\|} & \frac{-(Y^i - Y_r)}{\|\mathbf{r}^s - \mathbf{r}_r\|} & \frac{-(Z^i - Z_r)}{\|\mathbf{r}^s - \mathbf{r}_r\|} \end{bmatrix} \quad (10)$$

225
 226 where $\mathbf{r}^s = (X^i, Y^i, Z^i)$ is the i th satellite position vector, $\mathbf{r}_r = (X_r, Y_r, Z_r)$ is the initial position
 227 vector of the receiver, n is the observed number of satellites.

228
 229 In the stage of parameter estimation, the stochastic model should be formed in the most accurate
 230 way to get optimum results. This issue requires more attention, especially in multi-GNSS
 231 applications. The variance-covariance matrix should represent actual stochastic conditions of
 232 observations. In the multi-GNSS, this situation can be outlined as the variance of the
 233 observations, depending on the GNSS system and the function of the satellite elevation angle,
 234 which can be written as (Pan et al., 2017):

$$\sigma^2 = \frac{\sigma_0^2}{\sin^2(El_e)} \quad (11)$$

236

237 where σ^2 is the variance value of the code observation, Ele is the satellite elevation angle, σ_0^2
 238 is the a priori variance of the related navigation system. Unlike GPS, Galileo, and BDS, which
 239 use Code Division Multiple Access (CDMA) signals, GLONASS uses Frequency Division
 240 Multiple Access (FDMA) signals, except for the recently modernized ones. It should be noted
 241 that different hardware delays exist on GLONASS receiver channels, as different frequencies
 242 are generated for each satellite. Therefore, GLONASS code observations contain inter-
 243 frequency biases (Wanninger, 2012).

244
 245 Since it is usually not desirable to determine these values for each satellite, this term is neglected
 246 in the stochastic model by assigning a higher variance value to the GLONASS code
 247 observations. As a result of the solution, the effect of this term is seen in the residuals of the
 248 GLONASS code observations (Cai and Gao, 2013). In addition, the BDS GEO satellites have
 249 lower performance than the BDS MEO and BDS IGSO satellites. The process should be
 250 advanced by reducing the weight of these satellites in the stochastic model (Zhang and Pan,
 251 2022). In the construction of the stochastic model, the chosen standard deviation (σ_0) of
 252 observations for GPS, GLONASS, Galileo, BDS-3 (MEO/IGSO), and BDS-3 (GEO) were
 253 taken as 0.3, 0.6, 0.3, 0.6, and 1.2 m, respectively.

254 255 2.2. Outlier Detection

256
 257 The receiver coordinates and clock offset parameters were estimated epoch-by-epoch using the
 258 Least Squares method without any constraints between observation epochs. It is necessary to
 259 pay attention to anomalous measurements in order to obtain optimal and reliable results. These
 260 measurements (i.e., outliers) may be seen in the dataset due to both hardware and environmental
 261 or external reasons. Thus, outliers should be identified and their impact on the solution should
 262 be minimized. Otherwise, undesired observations can degrade the positioning performance
 263 (Angrisano et al., 2020). The median absolute deviation (MAD) estimator is adopted because
 264 it is highly resistant to the outliers with a 50% breakdown point. The MAD estimator is
 265 formulated as follows (Rousseeuw and Leroy, 1987; Hekimoglu et al., 2014).

$$266 \quad MAD = 1.4826 \times median\{|l_1 - med|, |l_2 - med|, \dots, |l_n - med|\} \quad (12)$$

267

$$med = median\{l_1, l_2, \dots, l_n\} \quad (13)$$

268
 269 where l_i is the value of the i th observation in the measurement vector (I), n is the length of
 270 the vector. The $k \times MAD$ threshold is compared with each $(|l - med|)$ value. If a value is greater
 271 than the threshold value, the observation is marked as an outlier and excluded from the
 272 evaluation. Herein, the k value was set to 3. MAD represents the standard deviation for
 273 normally distributed data. With the 3σ edit rule, which is generally called the Hampel
 274 identifier, outliers can be detected (Pearson, 2001; Chiang et al., 2003). It should be pointed out
 275 that the approach performs inaccurately in multi-GNSS processing since different systems have
 276 different receiver clock offsets. Therefore, this method can be used to detect gross errors in pre-
 277 analyzing each system independently. As a result, it has functioned very efficiently in epochs
 278 where the degrees of freedom are low and outliers cannot be detected by common statistical
 279 methods.

280
 281 Following the processing stage of gross errors detection, for SPP results, an appropriate weight
 282 model is a crucial requirement. A robust weighting method was implemented to mitigate the
 283 impact of outliers that cannot be detected with the MAD technique in this study. The method
 284 creates an equivalent weighting matrix according to certain conditions by using the information
 285 obtained from the least-squares solution. The IGG (Institute of Geodesy and Geophysics)-III
 286 function was employed to calculate the matrix. The weighting of observations was adjusted
 287 with IGG-III as follows (Y. Yang et al., 2002; Guo and Zhang, 2014).

$$\gamma_i = \begin{cases} 1 & |\tilde{v}_i| \leq k_0 \\ \frac{k_0}{|\tilde{v}_i|} \left(\frac{k_1 - |\tilde{v}_i|}{k_1 - k_0} \right)^2 & k_0 < |\tilde{v}_i| \leq k_1 \\ 0 & |\tilde{v}_i| > k_1 \end{cases} \quad (14)$$

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Değiştirilmiş Alan Kodu

Değiştirilmiş Alan Kodu

Biçimlendirdi: Türkçe (Türkiye)

$$\bar{P}_i = P_i \gamma_{ii} \quad ; \quad \bar{P} = \begin{bmatrix} \frac{\gamma_{11}}{\sigma_1^2} & & & & \\ & \frac{\gamma_{22}}{\sigma_2^2} & & & \\ & & \ddots & & \\ & & & \ddots & \\ & & & & \frac{\gamma_{mm}}{\sigma_m^2} \end{bmatrix} \quad (15)$$

292
 293
 294 where \bar{P} is the equivalent weighting matrix, γ is the inflation factor, subscript i represents
 295 each observation, k_0 and k_1 are two threshold values and usually set $k_0 = 1.5 - 3$, $k_1 = 3 - 8$. \tilde{v}_i
 296 is the standardized residual, expressed by the following equation:
 297

$$\tilde{v}_i = \frac{v_i}{\sqrt{\hat{\sigma}_0^2 Q_{v_i}}} \quad (16)$$

298
 299 where $\hat{\sigma}_0^2$ is the estimate of unit weighted variance, Q_{v_i} is the cofactor matrix of residual.
 300

$$\hat{\sigma}_0^2 = \frac{\mathbf{v}^T \mathbf{P} \mathbf{v}}{n - u} \quad (17)$$

$$\mathbf{Q}_{v_i} = \mathbf{P}^{-1} - \mathbf{A} (\mathbf{A}^T \mathbf{P} \mathbf{A})^{-1} \mathbf{A}^T \quad (18)$$

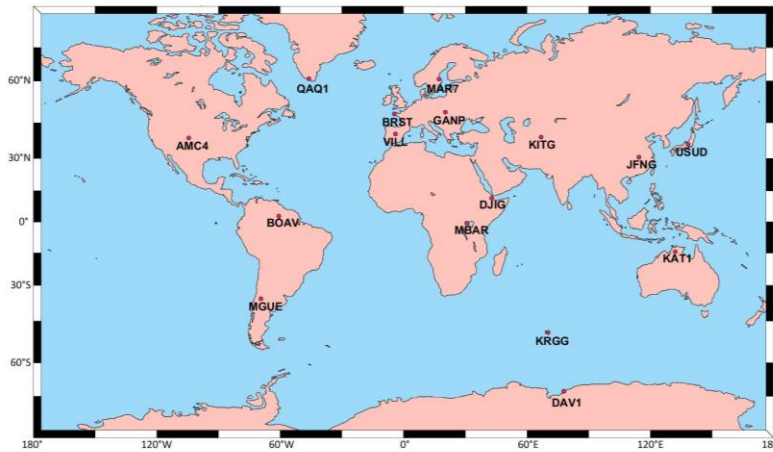
302
 303 where n is the number of observations, u is the number of unknowns in the equation. With the
 304 IGG-III function, the equivalent weighting matrix is constructed iteratively. The maximum
 305 standardized residual value of observations is taken into account. Equation (15) is used to
 306 reweight this observation. The processing steps are repeated with the new matrix. The
 307 procedure is repeated until there is no change in the standardized residual values obtained in
 308 the two iterations.

310 3. Experimental Design

311 3.1. Data Description

313
314 To investigate the SPP performances of all combinations of GPS, GLONASS, Galileo, and
315 BeiDou navigation systems (BDS)-3, 16 stations of the multi-GNSS experimental (MGEX)
316 monitoring network were selected. The 5-day observation files, between Day of Year (DOY)
317 137-141 (17-21 May 2021) with a sample interval of 30 seconds, were obtained from Crustal
318 Dynamics Data Information System (CDDIS) in Receiver Independent Exchange (RINEX)
319 format (available at: <https://cddis.nasa.gov/archive/gnss/>). The essential aspect in the selection
320 of these stations is throughout the world analyzing the performance of GNSS systems. The
321 geographic distribution of these MGEX stations is shown in Figure 1.

322



323
324 Figure 1. Geographic distribution of 16 selected IGS MGEX stations in this study

325

326 The main objective of this research is to assess the real-time positioning performance, using
327 pseudorange observations, with external orbit and clock products offered by international
328 GNSS Monitoring and Assessment System (iGMAS). iGMAS, which is similar to the
329 International GNSS Service (IGS) in terms of mission, supplies these products free of charge,
330 calculating them with its global monitoring station's data. [For more information about iGMAS's
331 tracking network, Analysis Centers, and mission, \(Zhu et al., 2022\) and \(Zhou et al., 2022\) can
332 be examined.](#) Both orbit and clock products are accessible with a sampling interval of 15
333 minutes. The ultra-rapid (so-called super fast) orbit and clock offset files provided by iGMAS
334 for 6-hourly include GPS, GLONASS, Galileo, and BDS (available at:
335 <http://www.igmas.org/Product/>). Thus, these products enable near-real-time and real-time

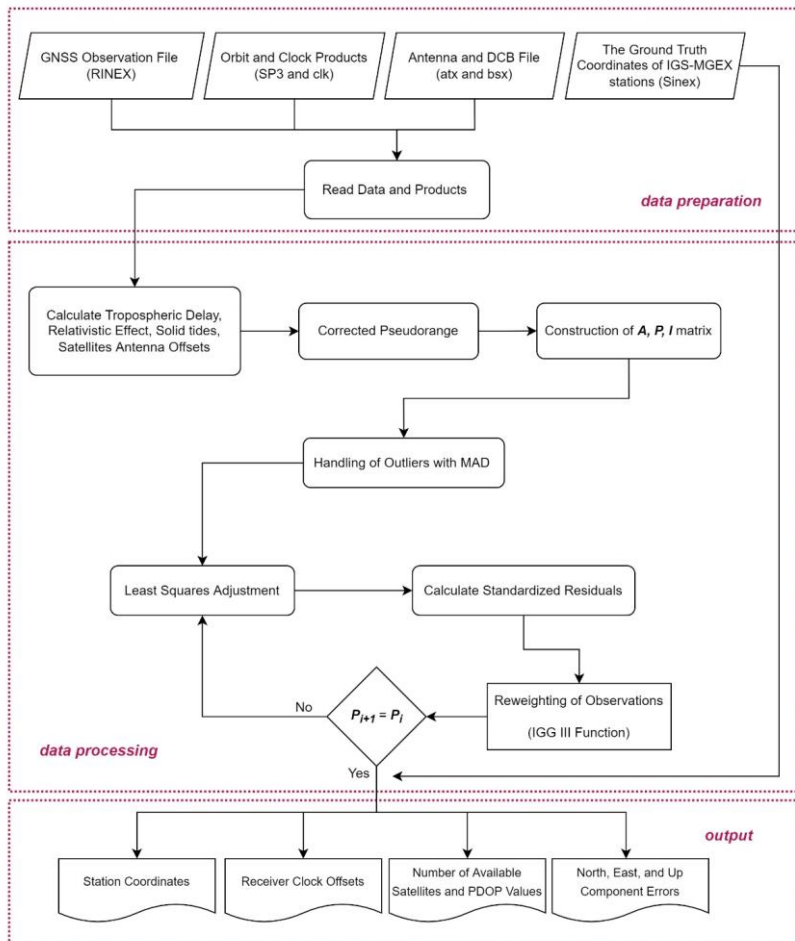
336 applications with quad constellation integration. During the processing stage, iGMAS products
337 were sorted through BDS System Time (BDT), whereas the observations referred to the GPS
338 Time (GPST). The time difference between BDT and GPST is 14 seconds (iGMAS, 2022). For
339 resolving this inconsistency, one should be getting them into alignment with the chosen
340 reference time (GPST) system. For the improvement of SPP results, the Differential Code Bias
341 (DCB) values should be applied to the pseudorange observations. The Bias Solution
342 Independent Exchange Format (Bias-SINEX) file contains the bias values, which vary
343 depending on the satellites and type of code measurements, provided by the Center for Orbit
344 Determination in Europe (CODE) on a monthly basis. In our study, C1C code observations at
345 GPS L1 frequency at BRST, GANP, KRGG, BOAV, MAR7, and JFNG stations, were turned
346 into C1W observations using differential code biases (C1C-C1W). The current antenna file
347 (igs14 2163.atx) was applied to convert the satellite coordinates to their phase centers.
348 Furthermore, the precise coordinates of the stations provided by IGS weekly were considered
349 as the ground truth. Accuracy performance analyses were carried out by comparing the ground
350 truth coordinates with the epoch-by-epoch coordinates acquired from the SPP solution. ▲

Biçimlendirdi: Türkçe (Türkiye)

352 3.2. Processing Strategies

353
354 Within the context of the study, Matlab-based in-house code has been developed to evaluate
355 the SPP solution of single, dual, triple, and quad combinations of GPS, GLONASS, Galileo,
356 and BDS-3 constellations. The flowchart of the code is shown in Figure 2.

357
358
359



360
 361 Figure 2. Flowchart of SPP processing code. The code includes three components: data
 362 preparation component, data processing component, and output component (SPP: Single Point
 363 Positioning, MAD: Mean Absolute Deviation, IGG-III: Institute of Geodesy and Geophysics-
 364 III, PDOP: Position Dilution of Precision)

365
 366 The orbit and clock offsets of BDS satellites are not included in the ultra-rapid products
 367 provided by IGS ACs except for WHU and CNES. However, iGMAS ultra-rapid products,
 368 which incorporate orbit and clock offsets of BDS-3 satellites, give the advantage of a solution
 369 for real-time applications with the integration of the quad constellation. In addition, the
 370 performance of BDS-3 with both single and other navigation system combinations can also be

371 evaluated. Furthermore, the developed program is capable of performing SPP solutions with
 372 both iGMAS and IGS products. The same observation data was analyzed again using the
 373 GeoForschungs Zentrum (GFZ) rapid orbit and clock offsets in order to compare the accuracy
 374 performance of all combinations. The methodologies used in these evaluation stages are
 375 described in Table 1.

376

377 Table 1. Processing settings for developed SPP-based program

Item	Models and Strategies
Observation	<i>IF</i> combination of code observations for GPS L1 and L2, GLONASS G1 and G2, Galileo E1 and E5a, BDS-3 B1 and B3
Sampling rate	30 s
Elevation cutoff	7°
Satellite orbit and clock offsets	6-hourly (00, 06, 12, 18) iGMAS ultra-rapid (predicted part) products, GFZ rapid products
Estimator	Epoch-by-epoch Weighted Least Squares
Weight Scheme	Elevation dependent, standard deviation of constellations: GPS:0.30 m, GLONASS: 0.60 m, Galileo: 0.30m, BDS-3: (IGSO and MEO: 0.60 m, GEO: 1.20m)
Tropospheric Delay	UNB3m Tropospheric Model
Satellites PCO	igs14_2163.atx
DCB correction	Converted C1C to C1W for GPS, C1C to C1P for GLONASS using CODE monthly Bias product, Daily DCBs provided by Chinese Academy of Sciences during solution with rapid products
Solid earth tide, the relativistic effect	Corrected (Petit and Luzum, 2010)

378

379 4. Results

380

381 4.1. Assessment of iGMAS ultra-rapid orbit and clock offset products

382

383 The performance of iGMAS ultra-rapid (predicted part) products was compared with the Center
 384 for Orbit Determination in Europe (CODE) final precise products throughout the solution days.
 385 The orbit and clock offset accuracies of the available satellites are shown in Figure 3. While the
 386 RMS values were considered for orbital accuracies, Standard Deviation (STD) values were
 387 calculated for clock offsets due to the different clock datums of products. The average 3D orbit
 388 accuracy of available satellites was estimated as 0.07, 0.15, 0.13, and 0.25 m for GPS,
 389 GLONASS, Galileo, and BDS-3, respectively. BDS-3 is distinct from the others because of the
 390 low orbit accuracy of IGSO satellites. The clock offset STDs for GPS, GLONASS, Galileo, and
 391 BDS-3 were found as 2.03, 2.27, 1.92, and 1.85 ns, respectively. In addition, it should be
 392 pointed out that the situations with no orbit and clock offset information, no clock offset
 393 information while the orbit information is available, or vice versa, have been observed in

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iGMAS products for some satellites. Providing the products for all satellites in the future will increase the effectiveness of iGMAS in positioning.

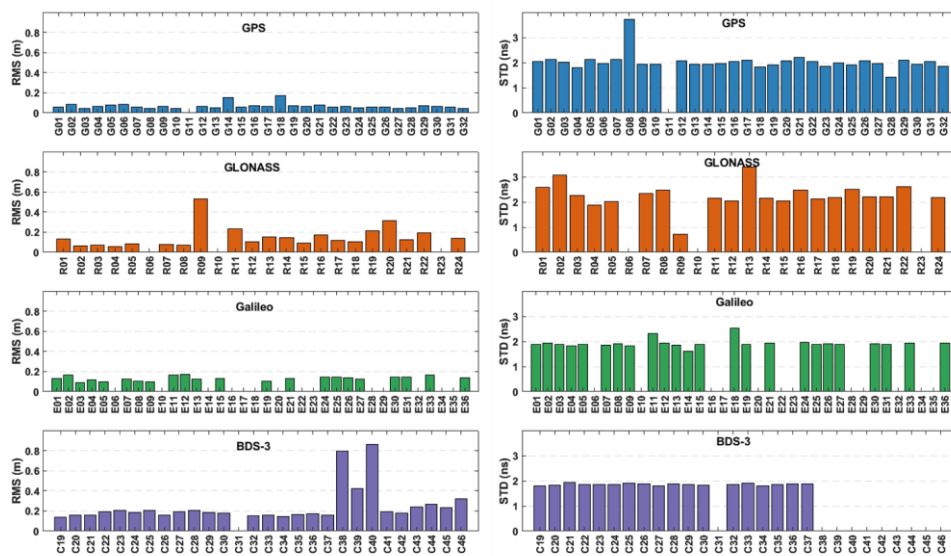


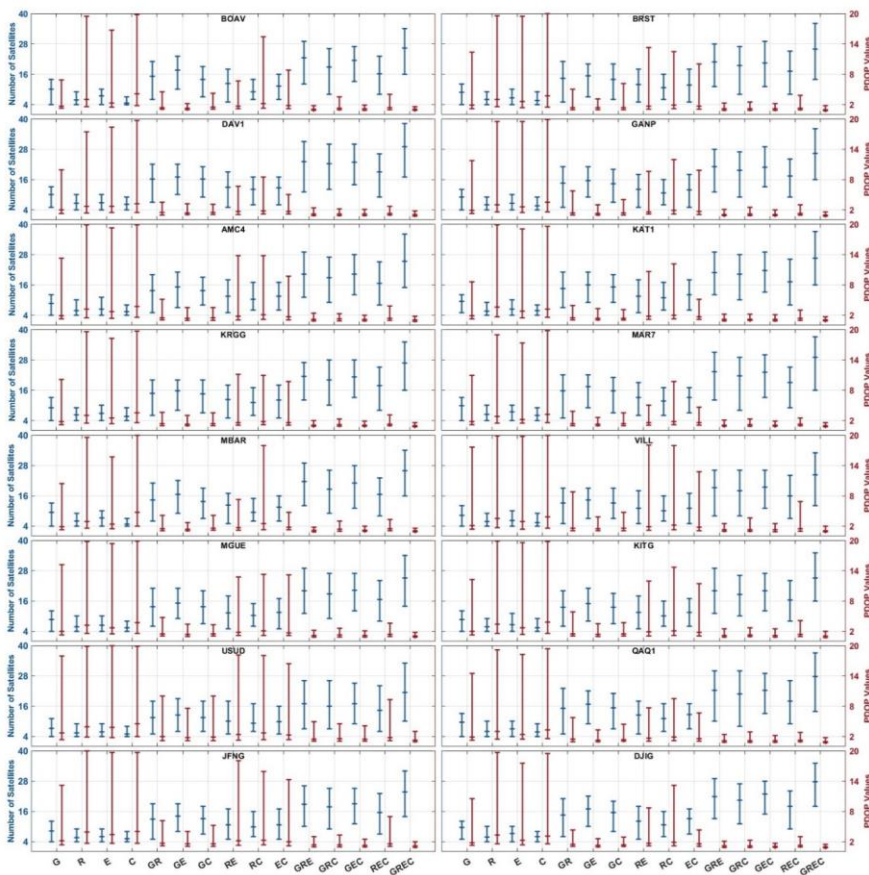
Figure 3. The average RMS values for the orbit and the average STD values for the clock offset according to the final products produced by IGS CODE for available satellites

4.12. Availability and PDOP Analyses

The number of the available satellite and Position Dilution of Precision (PDOP) values were obtained for all combinations of GPS, GLONASS, Galileo, and BDS-3 in SPP solutions for 16 IGS stations with iGMAS ultra-rapid products. Positioning performance was negatively affected by the satellite availability of iGMAS ultra-rapid products. To be more specific, the satellites associated with the observed session may not be present in the precise products. Therefore, the satellites not having clock offset and/or orbit information were ignored in the solution. In the developed code, the solution outcomes were examined according to various cases. In this context, the SPP solutions are possible for cases containing at least 4, 5, 6, and 7 satellites on epochs for single, dual, triple, and quad systems, respectively. Besides, if the PDOP value is less than 20 in any epoch, the solution is accepted as a valid result.

Figure 3-4 depicts the maximum, mean, and minimum number of satellites employed in the solution at the stations, as well as the PDOP values. The average number of satellites in GPS,

415 GLONASS, Galileo, and BDS-3 were 9-10, 5-6, 7-8, and 5-6, respectively, while the PDOPs
 416 of these systems were 1.81, 3.30, 2.30, and 3.11 respectively. The results showed that
 417 GLONASS and BDS-3 performed worse than other systems in terms of the minimum satellite
 418 and maximum PDOP criteria. It is an expected result that this issue affects the positioning
 419 performance of GLONASS and BDS-3, which will be explained in more detail in the next
 420 section. On the other hand, it was observed that Galileo was the most stable system after GPS
 421 in terms of PDOP values and the number of satellites.



422
 423 Figure 34. Maximum, average, and minimum number of satellites and PDOP values for all
 424 combinations

425
 426 Within the dual constellations, on average, the PDOP values ranged between 1.29-and 1.91.
 427 Thus, dual-system solutions provided significant advantages in positioning accuracy compared

428 with single-systems. However, it should be noted that GLONASS- BDS-3 integration results
429 are not very compatible with the other dual constellations. Redundancy in SPP solutions for a
430 single epoch increased dramatically with an average of 17-23 and 27-28 visible satellites in all
431 triple and quad constellations integrations. The PDOP values were at an extremely ideal level
432 in these combinations. Furthermore, the narrower range of maximum and minimum PDOP
433 values indicates more reliable positioning. The positioning accuracy, reliability, and
434 redundancy provided by the quad constellation were especially remarkable.

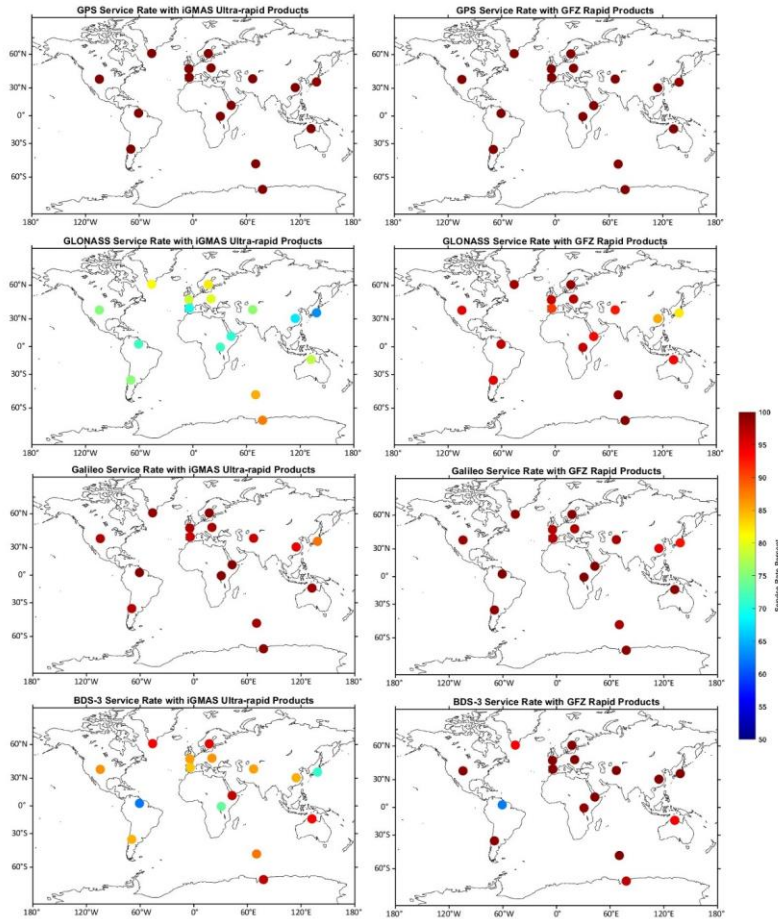
435

436 4.23. Single-System SPP Performance

437

438 The positioning accuracy of navigation systems was assessed for the solution of single system
439 cases (GPS-only, GLONASS-only, Galileo-only, and BDS-3-only) using the developed SPP
440 code. However, as indicated in the previous section, the number of visible satellites is a critical
441 condition for all positioning techniques, not just SPP. When there are insufficient satellites, or
442 even when there are enough satellites, the orbit-clock products of some of these satellites are
443 unavailable, and position acquisition is impossible. Theoretically, in single system situations,
444 orbit and clock information of at least four satellites must be available so that a solution can be
445 obtained. However, due to satellite's health or other factors, some satellite's orbit and clock
446 products may not be available. This circumstance particularly occurs in ultra-rapid (in the
447 predicted-part) products. For this reason, some epochs could not be solved and were marked
448 as unresolved. In other words, the success of providing solutions throughout all epochs is
449 referred to as the service rate for navigation systems, and it is defined by valid solutions. Figure
450 4-5 shows the average service rate of independent navigation systems in 5-day solutions
451 employing both iGMAS ultra-rapid and GFZ rapid products. The results reveal that the service
452 rates of GPS and Galileo are very similar. The GPS service rate for both products is more than
453 98% for all stations. Ultra-rapid products in Galileo solutions have a service rate of more than
454 94% at all stations, except for the USUD station, which has an 88% service rate. GLONASS
455 had the worst service rate among all navigation systems. While the average GLONASS service
456 rate for iGMAS products was 76%, the service rate for all stations increased with GFZ products
457 and reached an average of 95%. Furthermore, the GLONASS service rate is slightly better in
458 the regions close to the pole due to its high orbital inclination. Except for the BOAV station,
459 the BDS-3 showed an average service rate performance of 87% with ultra-rapid products.
460 Excluding the same station as the GFZ rapid products, the service rate has exceeded 98%. The
461 service rate of this station has not changed and has remained at 62%. This situation is considered

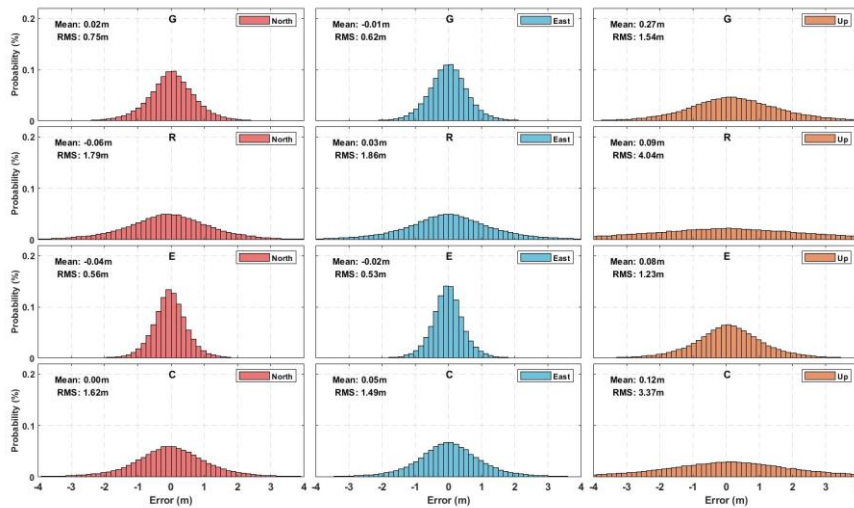
462 to be caused by geographical location. When the service rates for GLONASS and BDS-3 are
 463 assessed, it is clear that the quality of iGMAS ultra-rapid products for the systems should be
 464 upgraded.



465
 466 Figure 45. Service rate achieved by iGMAS ultra-rapid and GFZ rapid products in the event of
 467 a GPS, GLONASS, Galileo, and BDS-3 single-system for 16 MGEX stations

468
 469 Focusing on the individual GNSS SPP (GPS: G, GLONASS: R, Galileo: E, BDS-3: C)
 470 performance, Figure 5-6 shows the distribution of positioning errors for the north, east, and up
 471 components for 16 stations. The distributions, containing a total of 230400 epoch (the 5-day)
 472 SPP solutions, are expressed as probability percentages. In addition, the average error and root
 473 mean square (RMS) statistics were obtained for each component (Figure 56). When the RMS

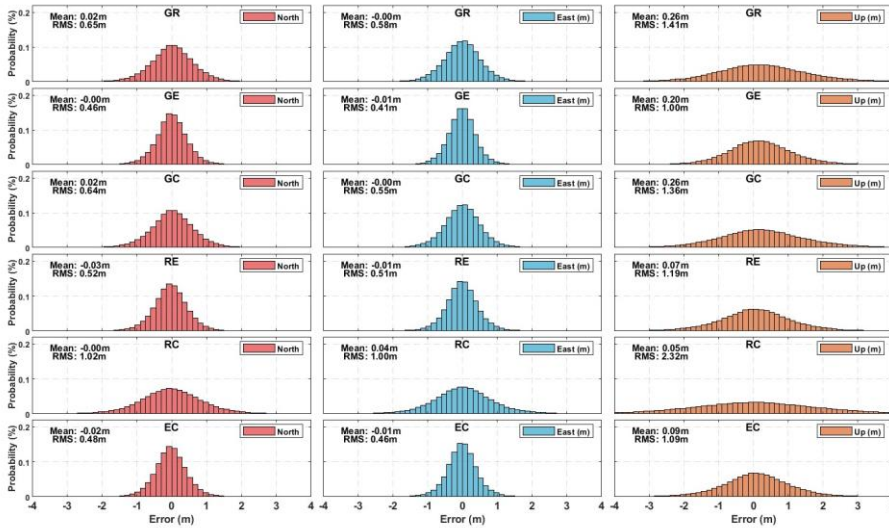
474 values for the north, east, and up components of the generated error distributions are analyzed
475 in detail, Galileo emerges as the single-system with the best positioning performance, with the
476 RMS values of 0.56, 0.53, and 1.23 m, respectively. All of Galileo's error components have
477 lower RMS values than the others. Thus, it can be said that the quality of the Galileo signals
478 used makes a significant contribution to the positioning accuracy. This result is in agreement
479 with the findings reported in a study with multi-GNSS SPP (Zhang and Pan, 2022). Then, GPS
480 showed higher positioning accuracy than GLONASS and BDS-3 with the RMS statistic being
481 0.75, 0.62, and 1.54 m in the north, east, and up components, respectively. GLONASS had the
482 worst single-system positioning performance with the RMS statistics of 1.79, 1.86, and 4.04 m
483 in the three components, respectively. BDS-3 offered better positioning performance than
484 GLONASS, with the RMS statistics for three components being 1.62, 1.49, and 3.37 m,
485 respectively. GPS and Galileo outperformed the positioning accuracy of GLONASS and BDS-
486 3. Although BDS-3 can compete with GPS and Galileo in terms of service rate and PDOP
487 values with iGMAS ultra-rapid products, it performed poorly for positioning accuracy. In the
488 mean error, the horizontal components have a maximum error of 7 cm and the vertical
489 component has a maximum error of 27 cm. These mean errors, especially in single solutions,
490 provide evidence that systematic errors are well modeled. It should be also noted that in a single
491 system solution, low redundancy might result in a potential outlier going undetected, causing
492 problems with the solutions. This occurred, particularly in some GLONASS and BDS-3
493 solutions. The Median Absolute Deviation (MAD) method prevented this issue, whereas the
494 IGG-III robust approach did not be operated properly in epochs with low redundancy.
495



496
 497 Figure 56. Error distributions of epoch-by-epoch SPP solutions in the north, east, and up
 498 components with iGMAS ultra-rapid products for GPS-only, GLONASS-only, Galileo-only,
 499 and BDS-3-only

500
 501 4.34. Dual-System SPP Performance

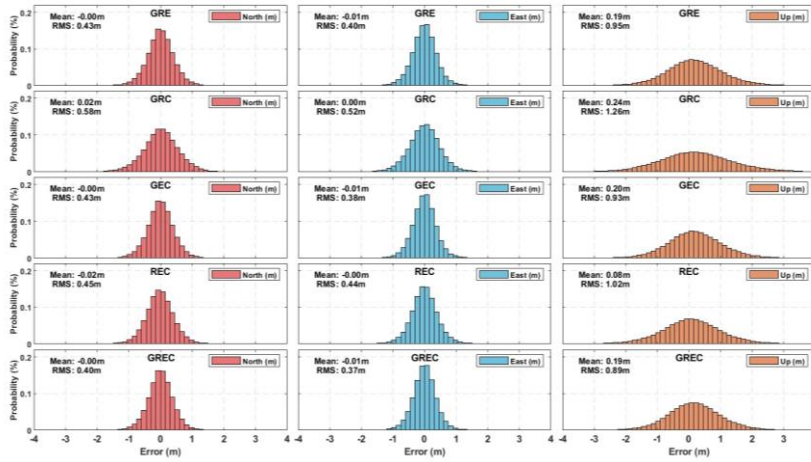
502
 503 Figure 6-7 demonstrates the distributions of positioning errors, RMS statistics, and mean
 504 statistics for all dual combinations of the navigation systems utilized in the study. First of all,
 505 dual constellation combination solutions considerably improved the positioning accuracy as
 506 compared with single-GNSS systems. The best dual combination performance was produced
 507 with the GPS/Galileo solution. RMS statistics of the north, east, and up components were 0.46,
 508 0.41, and 1.00 m, respectively. Furthermore, as a significant finding, the Galileo-based
 509 combinations outperformed the GPS-based combinations in terms of all metrics. Considering
 510 the GLONASS and BDS-3 combinations of GPS and Galileo, BDS-3 further improved
 511 positioning accuracy according to GLONASS. BDS-3 enhanced the three-dimensional (3D)
 512 RMS errors of GPS-only, Galileo-only, and GLONASS-only solutions by about 12%, 12%, and
 513 44%, respectively. Among the dual combinations, GLONASS-BDS-3 had the worst outcomes.
 514 Also, combining BDS-3 and GLONASS did not yield the same quality results as using GPS
 515 and Galileo alone.



516
 517 Figure 67. Error distributions of epoch-by-epoch SPP solutions in the north, east, and up
 518 components with iGMAS ultra-rapid products for GPS/GLONASS, GPS/Galileo, GPS/BDS-3,
 519 GLONASS/Galileo, GLONASS/BDS-3, Galileo/BDS-3

520
 521 4.45. Triple-System and Quad-System SPP Performance
 522

523 The error distributions, RMS, and error values for triple and quad combinations are shown in
 524 Figure 78. The quad combination produced higher accurate results than all the combinations
 525 from the 5-day epoch-by-epoch solution of the 16 stations dataset. The RMS values of 0.54 m
 526 horizontally and 0.89 m vertically were estimated for the quad constellation. The benefit of
 527 employing the quad constellation in real-time applications, with its RMS value of 1.02 m in 3D,
 528 should be highlighted. Besides, it can be noted that there is no significant difference between
 529 the quad constellation results and the GPS/GLONASS/Galileo and GPS/Galileo/BDS-3 results.
 530 Because they have almost nearly the same accuracy as quad-constellation, these triple
 531 combinations can be employed in instances where quad combinations are not attainable.
 532 However, the accuracy of the GLONASS/Galileo/BDS-3 was better than the
 533 GPS/GLONASS/BDS-3 results. When GLONASS and BDS-3 were used simultaneously in the
 534 triple combination, the accuracy was considerably superior to that of its dual combination.



535

536

537 Figure 78. Error distributions of epoch-by-epoch SPP solutions in the north, east, and up
 538 components with iGMAS ultra-rapid products for GPS/GLONASS/Galileo,
 539 GPS/GLONASS/BDS-3/, GPS/Galileo/BDS-3, GLONASS/Galileo/BDS-3, and
 540 GPS/Galileo/GLONASS/BDS-3

541

542 4.56. iGMAS ultra-rapid products versus GFZ Rapid Products

543

544 SPP solutions were performed with the same dataset utilizing also GFZ rapid orbit and clock
 545 products to assess the quality of iGMAS ultra-rapid orbit and clock products. Table 2 gives the
 546 5-day mean RMS statistics for all combinations of each station based on both GFZ and iGMAS
 547 products. The most important point to highlight is that while the GLONASS-only and BDS-3-
 548 only results in the two different solutions differed dramatically, the GPS-only and Galileo-only
 549 results did not differ significantly. Additionally, the performance of GLONASS/BDS-3
 550 combined improved by 35% with GFZ products. However, when GPS/Galileo was taken into
 551 consideration, it showed by 9% improvement. It can be concluded that GLONASS and BDS-3
 552 orbit and clock products produced by iGMAS lagged behind GPS and Galileo in terms of
 553 availability and accuracy. The positioning performance of the systems, both individually and in
 554 combination with other systems, is predicted to improve if the orbit and clock products of BDS-
 555 3 and GLONASS are enhanced. In triple-system solutions, it was observed that the accuracies
 556 obtained with GFZ and iGMAS products were consistent with each other. Nevertheless, the
 557 GRC combination including GLONASS and BDS-3 produced poorer results. Finally, the 3D

558 accuracy obtained with real-time positioning in the quad constellation integration was only 0.13
 559 m less than the result obtained with the rapid products. Namely, precise navigation needs can
 560 be met using the combination of GREC with the ultra-rapid products provided for GPS,
 561 GLONASS, Galileo, and BDS-3.

562
 563 Table 2. 5-day average 3D RMS statistics based on iGMAS ultra-rapid and GFZ rapid products
 564 for all stations (unit: m)

3D RMS for iGMAS Ultra-Rapid Products															
Station	G	R	E	C	GR	GE	GC	RE	RC	EC	GRE	GRC	GEC	REC	GREC
BOAV	2.57	5.76	1.72	7.88	2.34	1.61	2.42	1.63	4.02	1.63	1.53	2.20	1.54	1.51	1.47
BRST	2.26	4.66	1.67	3.88	1.98	1.41	1.93	1.52	2.72	1.36	1.29	1.74	1.29	1.27	1.20
DAV1	1.71	3.57	1.44	3.46	1.47	1.10	1.46	1.29	2.36	1.28	1.01	1.31	1.01	1.15	0.95
GANP	1.87	3.95	1.17	2.50	1.69	1.19	1.62	1.17	2.03	0.95	1.12	1.48	1.09	0.94	1.03
AMC4	1.65	3.97	1.33	3.97	1.55	1.09	1.48	1.36	2.42	1.17	1.05	1.38	1.02	1.12	0.98
KAT1	1.81	5.79	1.80	4.20	1.63	1.23	1.58	1.55	2.63	1.46	1.16	1.45	1.14	1.32	1.08
KRGG	1.70	4.32	1.25	2.93	1.52	1.07	1.48	1.22	2.07	1.08	1.01	1.36	1.00	1.03	0.95
MAR7	1.99	3.94	1.58	3.32	1.82	1.32	1.75	1.51	2.48	1.39	1.27	1.64	1.24	1.33	1.20
MBAR	1.23	5.88	1.20	4.70	1.22	0.84	1.10	1.27	3.43	1.11	0.84	1.10	0.80	1.12	0.80
VILL	1.80	5.10	1.56	4.59	1.65	1.18	1.59	1.47	2.86	1.42	1.13	1.49	1.11	1.27	1.07
MGUE	1.50	3.94	1.16	3.54	1.39	0.94	1.28	1.15	2.21	1.04	0.89	1.21	0.86	0.99	0.83
KITG	1.30	4.65	1.16	3.54	1.25	0.86	1.20	1.29	2.36	1.07	0.85	1.15	0.83	1.06	0.81
USUD	2.06	6.24	2.03	5.71	1.95	1.45	1.86	1.97	3.84	1.83	1.40	1.77	1.36	1.65	1.32
QAQ1	1.54	4.58	0.92	2.80	1.35	0.84	1.25	1.05	2.35	0.84	0.82	1.17	0.78	0.88	0.77
JFNG	2.30	5.48	1.83	3.52	2.07	1.52	1.95	1.62	2.64	1.41	1.42	1.78	1.36	1.31	1.28
DJIG	1.08	4.28	0.80	2.67	1.05	0.69	0.96	0.92	2.08	0.78	0.68	0.94	0.66	0.80	0.65
Mean	1.77	4.76	1.41	3.95	1.62	1.15	1.56	1.37	2.66	1.24	1.09	1.45	1.07	1.17	1.02
3D RMS for GFZ Rapid Products															
Station	G	R	E	C	GR	GE	GC	RE	RC	EC	GRE	GRC	GEC	REC	GREC
BOAV	2.52	3.18	1.59	7.88	2.15	1.52	2.36	1.38	2.66	1.52	1.40	2.03	1.47	1.32	1.35
BRST	2.26	3.36	1.62	2.39	1.88	1.41	1.81	1.34	1.71	1.25	1.27	1.56	1.25	1.11	1.14
DAV1	1.48	1.76	1.26	3.35	1.17	0.95	1.31	0.98	1.50	1.11	0.84	1.08	0.89	0.91	0.80
GANP	1.89	2.67	1.10	1.21	1.62	1.19	1.43	0.94	1.10	0.79	1.07	1.26	0.98	0.74	0.90
AMC4	1.50	2.43	1.23	2.00	1.33	0.98	1.28	1.07	1.52	1.05	0.93	1.16	0.91	0.94	0.86
KAT1	1.62	3.43	1.51	4.19	1.39	1.08	1.45	1.25	2.06	1.31	0.99	1.27	1.02	1.14	0.95
KRGG	1.63	2.40	1.11	1.73	1.37	1.00	1.29	0.99	1.25	0.87	0.91	1.12	0.88	0.80	0.81
MAR7	2.00	2.92	1.51	2.48	1.69	1.32	1.68	1.33	1.85	1.29	1.21	1.48	1.22	1.17	1.13
MBAR	1.09	3.74	1.08	2.65	1.04	0.78	0.96	1.06	1.83	0.97	0.76	0.91	0.73	0.92	0.71
VILL	1.56	3.20	1.38	2.25	1.39	1.05	1.32	1.19	1.74	1.14	0.97	1.19	0.95	1.02	0.89
MGUE	1.19	2.63	0.98	1.72	1.06	0.76	0.99	0.91	1.41	0.84	0.72	0.92	0.69	0.79	0.66
KITG	1.16	2.67	0.90	1.78	1.05	0.73	1.00	0.88	1.46	0.83	0.70	0.93	0.69	0.79	0.67
USUD	1.76	4.60	1.74	2.79	1.60	1.24	1.49	1.59	2.27	1.45	1.18	1.40	1.16	1.36	1.12
QAQ1	1.18	3.93	0.83	2.79	1.13	0.71	1.01	1.03	2.32	0.78	0.71	0.99	0.66	0.86	0.66
JFNG	2.28	4.60	1.66	1.85	1.97	1.45	1.64	1.41	1.56	1.16	1.32	1.47	1.17	1.06	1.09
DJIG	0.90	3.22	0.65	1.51	0.84	0.56	0.82	0.70	1.34	0.67	0.55	0.78	0.56	0.67	0.55
Mean	1.63	3.17	1.26	2.66	1.42	1.05	1.37	1.13	1.72	1.06	0.97	1.22	0.95	0.98	0.89
Mean RMS values difference (iGMAS-GFZ)															
	0.15	1.59	0.15	1.29	0.20	0.10	0.19	0.25	0.93	0.17	0.12	0.23	0.12	0.20	0.13
3D-RMS for iGMAS Ultra-Rapid Products															
Station	G	R	E	C	GR	GE	GC	RE	RC	EC	GRE	GRC	GEC	REC	GREC
BOAV	2.57	5.76	1.72	7.88	2.34	1.61	2.42	1.63	4.02	1.63	1.53	2.20	1.54	1.51	1.47
BRST	2.26	4.66	1.67	3.88	1.98	1.41	1.93	1.52	2.72	1.36	1.29	1.74	1.29	1.27	1.20
DAV1	1.71	3.57	1.44	3.46	1.47	1.10	1.46	1.29	2.36	1.28	1.01	1.31	1.01	1.15	0.95
GANP	1.87	3.95	1.17	2.50	1.69	1.19	1.62	1.17	2.03	0.95	1.12	1.48	1.09	0.94	1.03
AMC4	1.65	3.97	1.33	3.97	1.55	1.09	1.48	1.36	2.42	1.17	1.05	1.38	1.02	1.12	0.98
KAT1	1.81	5.79	1.80	4.20	1.63	1.23	1.58	1.55	2.63	1.46	1.16	1.45	1.14	1.32	1.08
KRGG	1.70	4.32	1.25	2.93	1.52	1.07	1.48	1.22	2.07	1.08	1.01	1.36	1.00	1.03	0.95

MAR7	1.99	3.94	1.58	3.32	1.82	1.32	1.75	1.51	2.48	1.39	1.27	1.64	1.24	1.33	1.20
MBAR	1.23	5.88	1.20	4.70	1.22	0.84	1.10	1.27	3.43	1.11	0.84	1.10	0.80	1.12	0.80
VILL	1.80	5.10	1.56	4.59	1.65	1.18	1.59	1.47	2.86	1.42	1.13	1.49	1.11	1.27	1.07
MGUE	1.50	3.94	1.16	3.54	1.39	0.94	1.28	1.15	2.21	1.04	0.89	1.21	0.86	0.99	0.83
KITG	1.30	4.65	1.16	3.54	1.25	0.86	1.20	1.29	2.36	1.07	0.85	1.15	0.83	1.06	0.81
USUD	2.06	6.24	2.03	5.71	1.95	1.45	1.86	1.97	3.84	1.83	1.40	1.77	1.36	1.65	1.32
QAQ1	1.54	4.58	0.92	2.80	1.35	0.84	1.25	1.05	2.35	0.84	0.82	1.17	0.78	0.88	0.77
JFNG	2.30	5.48	1.83	3.52	2.07	1.52	1.95	1.62	2.64	1.41	1.42	1.78	1.36	1.31	1.28
DJG	1.08	4.28	0.80	2.67	1.05	0.69	0.96	0.92	2.08	0.78	0.68	0.94	0.66	0.80	0.65
Mean	1.77	4.76	1.41	3.95	1.62	1.15	1.56	1.37	2.66	1.24	1.09	1.45	1.07	1.17	1.02

3D-RMS for GFZ Rapid Products

Station	G	R	E	C	GR	GE	GC	RE	RC	EC	GRE	GRC	GEC	REC	GREC
BOAV	2.52	3.18	1.59	7.88	2.15	1.52	2.36	1.38	2.66	1.52	1.40	2.03	1.47	1.32	1.35
BRST	2.26	3.36	1.62	2.39	1.88	1.41	1.81	1.34	1.71	1.25	1.27	1.56	1.25	1.11	1.14
DAV1	1.48	1.76	1.26	3.35	1.17	0.95	1.31	0.98	1.50	1.11	0.84	1.08	0.89	0.91	0.80
GANP	1.89	2.67	1.10	1.21	1.62	1.19	1.43	0.94	1.10	0.79	1.07	1.26	0.98	0.74	0.90
AMC4	1.50	2.43	1.23	2.00	1.33	0.98	1.28	1.07	1.52	1.05	0.93	1.16	0.91	0.94	0.86
KAT1	1.62	3.43	1.51	4.19	1.39	1.08	1.45	1.25	2.06	1.31	0.99	1.27	1.02	1.14	0.95
KRGG	1.63	2.40	1.11	1.73	1.37	1.00	1.29	0.99	1.25	0.87	0.91	1.12	0.88	0.80	0.81
MAR7	2.00	2.92	1.51	2.48	1.69	1.32	1.68	1.33	1.85	1.29	1.21	1.48	1.22	1.17	1.13
MBAR	1.09	3.74	1.08	2.65	1.04	0.78	0.96	1.06	1.83	0.97	0.76	0.91	0.73	0.92	0.71
VILL	1.56	3.20	1.38	2.25	1.39	1.05	1.32	1.19	1.74	1.14	0.97	1.19	0.95	1.02	0.89
MGUE	1.19	2.63	0.98	1.72	1.06	0.76	0.99	0.91	1.41	0.84	0.72	0.92	0.69	0.79	0.66
KITG	1.16	2.67	0.90	1.78	1.05	0.73	1.00	0.88	1.46	0.83	0.70	0.93	0.69	0.79	0.67
USUD	1.76	4.60	1.74	2.79	1.60	1.24	1.49	1.59	2.27	1.45	1.18	1.40	1.16	1.36	1.12
QAQ1	1.18	3.93	0.83	2.79	1.13	0.71	1.01	1.03	2.32	0.78	0.71	0.99	0.66	0.86	0.66
JFNG	2.28	4.60	1.66	1.85	1.97	1.45	1.64	1.41	1.56	1.16	1.32	1.47	1.17	1.06	1.09
DJG	0.90	3.22	0.65	1.51	0.84	0.56	0.82	0.70	1.34	0.67	0.55	0.78	0.56	0.67	0.55
Mean	1.63	3.17	1.26	2.66	1.42	1.05	1.37	1.13	1.72	1.06	0.97	1.22	0.95	0.98	0.89

565

566

567 5. Conclusion

568

569 With the advancement of GNSS technology, space-based positioning has become more
570 common in real-time applications particularly navigation, guiding, and surveying. Currently,
571 there are four independent global systems namely, GPS, GLONASS, Galileo, and BDS. For
572 precise positioning, many Analyze Centers (ACs) provide basic information about the satellites
573 as ultra-rapid products to real-time users get through the IGS MGEX network. Parallel to IGS,
574 iGMAS offers ultra-rapid products for the four global constellations using its own network
575 computed by 12 ACs. This study aimed to assess the performance of iGMAS ultra-rapid
576 products in navigations problems. To the fulfillment of the objective, SPP solutions were
577 performed for all 15 combinations (single, dual, triple, and quad) of the constellations using a
578 5-day dataset of 16 MGEX stations. In this context, an in-house code was developed in
579 MATLAB for SPP solutions. The MAD approach was successful in removing gross errors in
580 SPP solutions. The approach also supported reweighting robust technique IGG-III performance.
581 Additionally, all datasets were also processed with GFZ rapid products to assess the availability
582 of the iGMAS ultra-rapid products, service rate, and positioning accuracy. In comparison to
583 GPS and Galileo, the results indicated that GLONASS and BDS-3 had poorer service rates.

584 This situation can be interpreted as being dependent on the absence of some satellites data in
585 ultra-rapid products. When using the ultra-rapid products against the rapid products, on average,
586 the service rate decreased from 95% to 76% for GLONASS and 97% to 86% for BDS-3.
587 Moreover, there were no noticeable degradations in service rates for GPS and Galileo when
588 employing ultra-rapid products.

589
590 For accuracy validation of the iGMAS products, the RMS values were calculated using all
591 epoch-wise solution results. The single-system solutions showed that Galileo produced the best
592 results with RMS values of 0.56, 0.53, and 1.23 m in the north, east, and up components,
593 respectively. The accuracy achieved with Galileo-only was even better than some dual
594 combinations. This can be explained by Galileo's observations being less sensitive to the
595 multipath effect, including less noise, and being less influenced by non-modelable errors. The
596 worst solutions were generated using the GLONASS with its RMS values of 1.79, 1.86, and
597 4.04 m. The dual constellation solutions demonstrated that combinations with Galileo produced
598 better results than the GR, GC, and RC solutions. Especially, the RC solution differed from the
599 other dual solutions negatively. In the triple constellation the other results, except for the GRC,
600 varied in the range 0.43-0.45 m, 0.38-0.44 m, and 0.93-1.02 m in the north, east, and up
601 components, respectively. The quad solution results had the lowest RMS values of 0.40, 0.37,
602 and 0.89 m in the three components. The combination results with respect to their RMS values
603 from the worst to the best can be listed as R, C, RC, G, GR, GC, GRC, E, RE, EC, REC, GE,
604 GRE, GEC, GREC. Results produced in this study indicated that Galileo and its combinations
605 exhibited remarkable performance.

606
607 Finally, the accuracy level obtained in triple and quad combination (approximately 0.65 m
608 horizontal, 1 m vertical component) with the proposed algorithm without using any
609 augmentation systems can meet the requirements of many applications such as civil aviation,
610 smart agriculture practices, ship navigation, and pedestrian and vehicle tracking, autonomous
611 systems like Unmanned Aerial Vehicles (UAV), and for some road and railway applications.
612 the results showed that multi-GNSS navigation solutions with the iGMAS products can be used
613 in many areas requiring sub-meter accuracies including open sea navigations, oceanographic
614 surveying, drone positioning, and Geographical Information Systems (GIS) data collections.
615 Positioning accuracy and reliability can be increased by expanding iGMAS' network and
616 enhancing the availability of ultra-rapid products. For future study, the developed SPP

617 algorithm is planned to be tested in harsh environments such as urban canyons and forest areas
618 with multi-GNSS.

619

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628

629 Data availability

630 The data used in this study are available on the CDDIS and iGMAS websites.

631

632 Conflict of interest

633 The authors declare that they have no competing interests.

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635 References

636

637 **Angrisano, A., Gaglione, S., Crocetto, N. and Vultaggio, M.** (2020). PANG-NAV: a tool for
638 processing GNSS measurements in SPP, including RAIM functionality. *GPS Solutions*, **24**(1),
639 19.

640 **Bahadur, B.** (2022). A study on the real-time code-based GNSS positioning with Android
641 smartphones. *Measurement*, **194**, 111078.

642 **Bahadur, B. and Nohutcu, M.** (2021). Real-time single-frequency multi-GNSS positioning
643 with ultra-rapid products. *Measurement Science and Technology*, **32**(1), 014003.

644 **Beutler, G., Rothacher, M., Schaer, S., Springer, T. A., Kouba, J. and Neilan, R. E.** (1999).
645 The International GPS Service (IGS): An interdisciplinary service in support of Earth sciences.
646 *Advances in Space Research*, **23**(4), 631–653.

647 **Beutler, Gerhard, Moore, A. W. and Mueller, I. I.** (2009). The international global
648 navigation satellite systems service (IGS): development and achievements. *Journal of Geodesy*,
649 **83**(3–4), 297–307.

650 **Bury, G., Sośnica, K., Zajdel, R. and Strugarek, D.** (2022). GLONASS precise orbit
651 determination with identification of malfunctioning spacecraft. *GPS Solutions*, **26**(2), 36.

652 **Cai, C. and Gao, Y.** (2013). Modeling and assessment of combined GPS/GLONASS precise
653 point positioning. *GPS Solutions*, **17**(2), 223–236.

654 **Cao, X., Kuang, K., Ge, Y., Shen, F., Zhang, S., and Li, J.** (2022). An efficient method for
655 undifferenced BDS-2/BDS-3 high-rate clock estimation. *GPS Solutions*, **26**(3), 66.

656 **Cerretto, G., Tavella, P., Lahaye, F., Mireault, Y. and Rovera, D.** (2012). Near real-time
657 comparison and monitoring of time scales with precise point positioning using nrcan ultra-rapid
658 products. *IEEE Transactions on Ultrasonics, Ferroelectrics and Frequency Control*, **59**(3), 545–
659 551.

660 **Chen, M., Yuan, H., Ma, J., Li, Z., Guang, W., Zhang, J., and Zhang, H.** (2022).
661 Performance Evaluation of Near Real-time GNSS PPP Time Transfer with IGS Products.
662 International Journal of Electrical and Computer Engineering Research, **2**(3), 1–7.

663 **Chiang, L. H., Pell, R. J., and Seasholtz, M. B.** (2003). Exploring process data with the use
664 of robust outlier detection algorithms. *Journal of Process Control*, **13**(5), 437–449.

665 **CSNO-TARC.** (2022). Test and Assessment Research Center of China Satellite Navigation
666 Office, Available at: <http://www.csno-tarc.cn/en/system/constellation> [accessed 4 February
667 2022].

668 **Dow, J. M., Neilan, R. E. and Rizos, C.** (2009). The International GNSS Service in a changing
669 landscape of Global Navigation Satellite Systems. *Journal of Geodesy*, **83**(3–4), 191–198.

670 **El-Mowafy, A., Deo, M. and Kubo, N.** (2017). Maintaining real-time precise point positioning
671 during outages of orbit and clock corrections. *GPS Solutions*, **21**(3), 937–947.

672 **Elsobeiey, M. and Al-Harbi, S.** (2016). Performance of real-time Precise Point Positioning
673 using IGS real-time service. *GPS Solutions*, **20**(3), 565–571.

674 **EUSPA.** (2022). European GNSS Service Centre, Available at: [https://www.gsc-](https://www.gsc-europa.eu/system-service-status/constellation-information)
675 [europa.eu/system-service-status/constellation-information](https://www.gsc-europa.eu/system-service-status/constellation-information) [accessed 5 February 2022].

676 **Ge, Y., Chen, S., Wu, T., Fan, C., Qin, W., Zhou, F. and Yang, X.** (2021). An analysis of
677 BDS-3 real-time PPP: Time transfer, positioning, and tropospheric delay retrieval.
678 *Measurement*, **172**(November 2020), 108871.

679 **Geng, T., Cheng, L., Xie, X., Liu, J., Li, Z., and Jiang, R.** (2022). GNSS real-time precise
680 point positioning with BDS-3 global short message communication devices. *Advances in Space*
681 *Research*, **70**(3), 576–586.

682 **Guo, F. and Zhang, X.** (2014). Adaptive robust Kalman filtering for precise point positioning.
683 *Measurement Science and Technology*, **25**(10), 105011.

684 **Hadaś, T. and Bosy, J.** (2015). IGS RTS precise orbits and clocks verification and quality
685 degradation over time. *GPS Solutions*, **19**, 93–105.

686 **Hadaś, T., Kaplon, J., Bosy, J., Sierny, J. and Wilgan, K.** (2013). Near-real-time regional
687 troposphere models for the GNSS precise point positioning technique. *Measurement Science*
688 *and Technology*, **24**(5), 055003.

689 **Hekimoglu, S., Erdogan, B., Soycan, M. and Durdag, U. M.** (2014). Univariate Approach
690 for Detecting Outliers in Geodetic Networks. *Journal of Surveying Engineering*, **140**(2),
691 04014006.

692 **iGMAS.** (2022). International GNSS Monitoring & Assessment System, Available at:
693 <http://www.igmas.org/> [accessed 5 February 2022].

694 **Jiang, W., Liu, M., Cai, B., Rizos, C., and Wang, J.** (2022). An accurate train positioning
695 method using tightly-coupled GPS + BDS PPP/IMU strategy. *GPS Solutions*, **26**(3), 67.

696 **Jiao, G. and Song, S.** (2022). High-Rate One-Hourly Updated Ultra-Rapid Multi-GNSS
697 Satellite Clock Offsets Estimation and Its Application in Real-Time Precise Point Positioning.
698 Remote Sensing, **14**(5), 1257.

699 **Jiao, G., Song, S., Ge, Y., Su, K. and Liu, Y.** (2019). Assessment of BeiDou-3 and Multi-
700 GNSS Precise Point Positioning Performance. *Sensors*, **19**(11), 2496.

701 **Kiliszek, D. and Kroszczyński, K.** (2020). Performance of the precise point positioning
702 method along with the development of GPS, GLONASS and Galileo systems. *Measurement*,
703 **164**, 108009.

704 **Koch, K. R.** (1999). *Parameter Estimation and Hypothesis Testing in Linear Models.* Springer,
705 Berlin Heidelberg New York.

706 **Kouba, J. and Héroux, P.** (2001). Precise Point Positioning Using IGS Orbit and Clock
707 Products. *GPS Solutions*, **5**(2), 12–28.

708 **Leandro, R. F., Langley, R. B. and Santos, M. C.** (2008). UNB3m_pack: a neutral
709 atmosphere delay package for radiometric space techniques. *GPS Solutions*, **12**(1), 65–70.

710 **Li, X., Zhang, X., Ren, X., Fritsche, M., Wickert, J. and Schuh, H.** (2015). Precise
711 positioning with current multi-constellation Global Navigation Satellite Systems: GPS,
712 GLONASS, Galileo and BeiDou. *Scientific Reports*, **5**(1), 8328.

713 **Liu, X., Jiang, W., Chen, H., Zhao, W., Huo, L., Huang, L. and Chen, Q.** (2019). An
714 analysis of inter-system biases in BDS/GPS precise point positioning. *GPS Solutions*, **23**(4),
715 116.

716 **Lou, Y., Zheng, F., Gu, S., Wang, C., Guo, H. and Feng, Y.** (2016). Multi-GNSS precise
717 point positioning with raw single-frequency and dual-frequency measurement models. *GPS*
718 *Solutions*, **20**(4), 849–862.

719 **Montenbruck, O., Steigenberger, P., Khachikyan, R., Weber, G., Langley, R., Mervart,**
720 **L. and Hugentobler, U.** (2014). IGS-MGEX: Preparing the Ground for Multi-Constellation
721 GNSS Science. *Inside GNSS*, **9**(1), 42–49.

722 **Montenbruck, O., Steigenberger, P., Prange, L., Deng, Z., Zhao, Q., Perosanz, F., Romero,**
723 **I., Noll, C., Stürze, A., Weber, G., Schmid, R., MacLeod, K. and Schaer, S.** (2017). The
724 Multi-GNSS Experiment (MGEX) of the International GNSS Service (IGS) – Achievements,
725 prospects and challenges. *Advances in Space Research*, **59**(7), 1671–1697.

726 **Niell, A. E.** (1996). Global mapping functions for the atmosphere delay at radio wavelengths.
727 *Journal of Geophysical Research: Solid Earth*, **101**(B2), 3227–3246.

728 **Ogutcu, S. and Farhan, H. T.** (2022). Assessment of the GNSS PPP performance using ultra-
729 rapid and rapid products from different analysis centres. *Survey Review*, **54**(382), 34–47.

730 **Pan, L., Cai, C., Santerre, R. and Zhang, X.** (2017). Performance evaluation of single-
731 frequency point positioning with GPS, GLONASS, BeiDou and Galileo. *Survey Review*,
732 **49**(354), 197–205.

733 **Pearson, R. K.** (2001). [Exploring process data. *Journal of Process Control*, **11**\(2\), 179–194.](#)

734 **Petit, G. and Luzum, B.** (2010). IERS Conventions 2010, IERS Technical Note 36, Frankfurt
735 am Main: Verlag des Bundesamts für Kartographie und Geodäsie, 179 pp., ISBN 3-89888-989-
736 6.

737 **Revnivykh, S., Bolkunov, A., Serdyukov, A. And Montenbruck, O.** (2017). GLONASS. In
738 P. J. G. Teunissen & O. Montenbruck (Eds.), *Springer Handbook of Global Navigation Satellite*
739 *Systems* (pp. 219–245). Springer International Publishing.

740 **Rousseeuw, P. J. and Leroy, A. M.** (1987). Robust Regression and Outlier Detection. In
741 *Journal of the Royal Statistical Society. Series A (Statistics in Society)* (Vol. 152, Issue 1). John
742 Wiley & Sons, Inc.

743 **Satirapod, C., Anonglekha, S., Choi, Y.-S. and Lee, H.-K.** (2011). Performance Assessment
744 of GPS-Sensed Precipitable Water Vapor using IGS Ultra-Rapid Orbits: A Preliminary Study
745 in Thailand. *Engineering Journal*, **15**(1), 1–8.

746 **Shen, P., Cheng, F., Lu, X., Xiao, X. and Ge, Y.** (2021). An Investigation of Precise Orbit
747 and Clock Products for BDS-3 from Different Analysis Centers. *Sensors*, **21**(5), 1596.

748 **Shi, J., Xu, C., Li, Y. and Gao, Y.** (2015). Impacts of real-time satellite clock errors on GPS
749 precise point positioning-based troposphere zenith delay estimation. *Journal of Geodesy*, **89**(8),
750 747–756.

751 **Shu, Y., Fang, R., Liu, Y., Ding, D., Qiao, L., Li, G. and Liu, J.** (2020). Precise coseismic
752 displacements from the GPS variometric approach using different precise products: Application
753 to the 2008 MW 7.9 Wenchuan earthquake. *Advances in Space Research*, **65**(10), 2360–2371.

754 **Wang, L., Li, Z., Ge, M., Neitzel, F., Wang, Z. and Yuan, H.** (2018). Validation and
755 Assessment of Multi-GNSS Real-Time Precise Point Positioning in Simulated Kinematic Mode
756 Using IGS Real-Time Service. *Remote Sensing*, **10**(2), 337.

757 **Wanninger, L.** (2012). Carrier-phase inter-frequency biases of GLONASS receivers. *Journal*
758 *of Geodesy*, **86**(2), 139–148.

759 **Yang, Y., Song, L. And Xu, T.** (2002). Robust estimator for correlated observations based on
760 bifactor equivalent weights. *Journal of Geodesy*, **76**(6–7), 353–358.

761 **Yang, Yuanxi, Gao, W., Guo, S., Mao, Y. and Yang, Y.** (2019). Introduction to BeiDou-3
762 navigation satellite system. *Navigation*, **66**(1), 7–18.

- 763 [Yigit, C. O., El-Mowafy, A., Bezioglu, M. ve Dindar, A. A. \(2020\). Investigating the effects](#)
764 [of ultra-rapid, rapid vs. final precise orbit and clock products on high-rate GNSS-PPP for](#)
765 [capturing dynamic displacements. Structural Engineering And Mechanics, 73\(4\), 427–436.](#)
- 766 **Zhang, Z. And Pan, L.** (2022). Current performance of open position service with almost fully
767 deployed multi-GNSS constellations: GPS, GLONASS, Galileo, BDS-2, and BDS-3. *Advances*
768 *in Space Research*, **69**(5), 1994–2019.
- 769 **Zhou, F., Dong, D., Li, P., Li, X. and Schuh, H.** (2019). Influence of stochastic modeling for
770 inter-system biases on multi-GNSS undifferenced and uncombined precise point positioning.
771 *GPS Solutions*, **23**(3), 59.
- 772 **Zhou, W., Cai, H., Chen, G., Jiao, W., He, Q. and Yang, Y.** (2022). Multi-GNSS Combined
773 Orbit and Clock Solutions at iGMAS. *Sensors*, **22**(2), 457.
- 774 [Zhu, Y., Zhang, Q., Mao, Y., Cui, X., Cai, C., and Zhang, R. \(2022\). Comprehensive](#)
775 [performance review of BDS-3 after one-year official operation. Advances in Space Research,](#)
776 [no. xxxx, Advance Access, published August 2022, 10.1016/j.asr.2022.08.020.](#)